

# An inhomogeneous stabilized node-based smoothed radial point interpolation method for the multi-physics coupling responses of functionally graded magneto-electro-elastic structures

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## ABSTRACT

In this paper, an inhomogeneous stabilized node-based smoothed radial point interpolation method (ISNS-RPIM) is developed for analyzing functionally graded magneto-electro-elastic (FGMEE) structures in hygrothermal environment. The change of material gradient follows the exponential function distribution. Combining with the gradient smoothing technique (GST) and the coupling constitutive relationship, and considering the hygrothermo-magneto-electro-elastic (HMEE) coupling effects of FGMEE material, the equations for ISNS-RPIM are deduced. The generalized displacement of the FGMEE structures is obtained, and the error in total energy norm, CPU time and efficiency of ISNS-RPIM, NS-RPIM and FEM are researched. Numerical examples are provided to explore the influence of the following parameters in FGMEE structures including exponential factor, hygrothermal loading and empirical constants, and demonstrate the correctness, high effectivity and insensitivity to mesh distortion of ISNS-RPIM. The present method has shown great potential in solving practical complex problems.

## 1. Introduction

As an inhomogeneous composite, functionally graded material (FGM) is utilized extensively in engineering applications: aerospace vehicles and electronic products, etc., the composition and properties of FGM could vary continuously along one or more specific directions [1–4]. Due to the property of reducing the stress concentration in layered composites, FGM has been widely concerned by researchers [5]. Scholars have made magneto-electro-elastic material (MEE) by combining piezomagnetic materials with piezoelectric materials. This material exhibits the unique ability in energy interconversion in multi-physics fields, which is used in a variety of energy harvesting systems, sensors, etc. [6–10]. Combining the merits of FGM and MEE material, FGMEE material is proposed to produce intelligent structures with higher performance. Recently, the multi-physics coupling analysis

(thermal environment, humidity, mechanical loads) for FGMEE material is a new topic due to the complexity of the working environment of intelligent devices.

The mechanical behavior of FGMEE structures has been deeply studied [11–14], among many algorithms, the finite element method (FEM) has been broadly developed [15,16]. Bhargale and Ganesan [17, 18] have calculated the natural frequency and static responses of the FGMEE plate under simply-supported boundary conditions and different exponential factors. Zenkour et al. [19–21] studied the infinite FGMEE hollow cylinder in thermal environment and calculated the radial displacement, temperature and stresses. Kattimani et al. [22,23] investigated the geometric nonlinear problems of traditional FGMEE plates using FEM. Then, they researched the porous FGMEE structures. The wave dispersion properties for nano-FGMEE beams were studied by Ebrahimi et al. [24], errors between the results and the analytical

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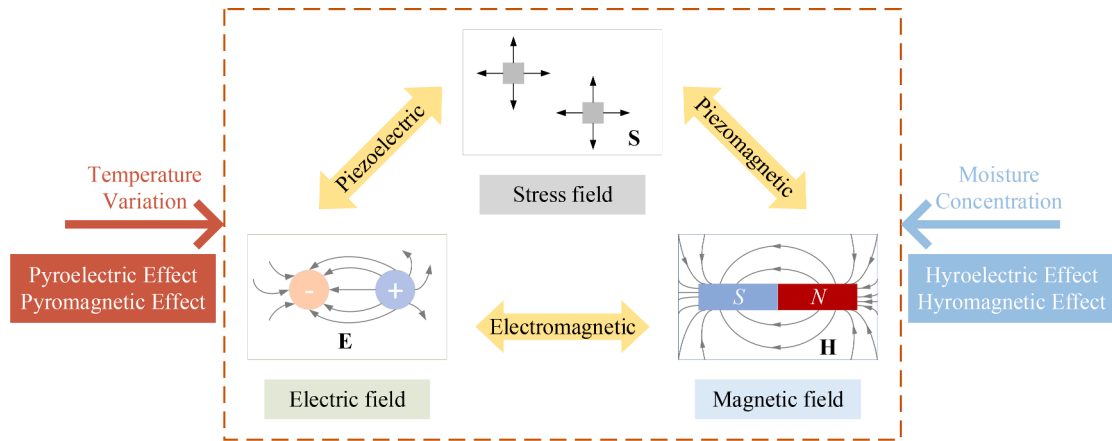


Fig. 1. Hygrothermo-magneto-electro-elastic coupling effects.

solutions were small. Mahesh et al. [25,26] researched the influence of MEE coupling effect in thermal environment on dynamic problems of FGME plates and carbon reinforced FGME structures, respectively. Recently, Vinyas [27] analyzed the dynamics of porous annular and circular FGME plates and obtained the factors affecting the frequency of the structures.

Studies have shown that hygrothermal environment could make smart structures more sensitive and FGME materials under hygrothermal loadings could produce various multi-physics coupling effects (Fig. 1) [28]. Vinyas et al. [29–31] considered the influence of thermal and hygroscopic field effects and investigated the response of MEE plates using FEM. Then, the research was extended to the skew MEE plate [32]. Dai et al. [5] obtained the responses of porous FGME annular plate subject to hygrothermal loadings. FGME nano-plate's damped frequency was calculated by Jafari et al. [33]. Saadatfar [1] considered the creep strain and analyzed the FGME hollow cylinders in hygrothermal environment under plane strain conditions and obtained the generalized displacement. Nguyen-Thoi et al. [34] investigated a numerical method for solving porous FGME plates about the free vibration and optimizing control problems.

With the constant advancement of numerical approaches, researchers devoted themselves to developing various algorithms to solve complex problems more accurately. Liu [35] proposed a series of smoothed FEMs to efficiently solve acoustic problems [36,37], multi-physics coupling problems [38], etc. Liu and Zhang [39] combined the generalized GST and weakened weak ( $W^2$ ) form with RPIM to propose the smoothed RPIM (S-RPIM) [39,40], whose governing equations are without independent variable's derivative. The  $W^2$  form without the derivative terms reduces the requirement of the approximation function [41]. In this article, the multi-physics coupling problems are solved using node-based S-RPIM (NS-RPIM) whose smoothing area is based on field nodes [42–44]. The essential boundary condition is easy to apply owing to the RPIM shape function satisfying the properties of Kronecker Delta functions. About strain energy, NS-RPIM has the advantage of upper bound property over FEM. Coordinate transformation and mapping are not required is another feature of NS-RPIM, thus NS-RPIM performs well in dealing with mesh distortion problems [45–47]. However, models constructed based on node-based smoothed numerical methods are 'overly-soft', this property could result in temporal instability in free vibration analysis or dynamic analysis [39,48]. Recently, stable node-based FEM was proposed to solve practical engineering problems including fracture analysis [49], acoustic problems [50,51], multi-physics coupling problems [48,52,53], elasticity problems [54,55], etc. Due to the stable terms, the node-based smoothed models' non-zero energy modes were eliminated.

In the previous work [52,56], stabilized NS-RPIM (SNS-RPIM) was applied to investigate the FGME and MEE material in the presence of

temperature conditions, then we investigated the factors affecting the responses of MEE materials under hygrothermal loadings. In this work, we proposed an ISNS-RPIM to accurately study the inhomogeneous HMEE coupling effects. Considering the HMEE coupling effect, the constitutive equations were derived. The node-based smoothing domain was established by triangular cells, then the domain was closed to a circular region with equal area. Four supplementary integration points are selected to derive the stable terms. This stabilization operation does not include any uncertain parameters, which is very simple. The numerical examples demonstrated the characteristics of ISNS-RPIM including temporal stability, high accuracy and efficiency for analyzing FGME materials. The main influencing factors of the FGME structure are analyzed, including the exponential factor, temperature and moisture concentration variation, and empirical constants.

Structure of this paper: In Section 2, the basic theories of FGME materials are introduced. Section 3 represents ISNS-RPIM. Section 4 contains numerical examples using ISNS-RPIM. The last section contains our conclusions.

## 2. Basic equations of FGME materials

As for FGME materials, the coefficients are considered to vary constantly along  $z$ -direction. The expressions of the coefficients are

$$\mathbf{C}(\mathbf{x}) = \mathbf{C}^0 f(\mathbf{x}), \mathbf{e}(\mathbf{x}) = \mathbf{e}^0 f(\mathbf{x}), \mathbf{q}(\mathbf{x}) = \mathbf{q}^0 f(\mathbf{x}) \quad (1)$$

$$\boldsymbol{\varepsilon}(\mathbf{x}) = \boldsymbol{\varepsilon}^0 f(\mathbf{x}), \boldsymbol{\mu}(\mathbf{x}) = \boldsymbol{\mu}^0 f(\mathbf{x}), \mathbf{m}(\mathbf{x}) = \mathbf{m}^0 f(\mathbf{x}) \quad (2)$$

$$\boldsymbol{\beta}(\mathbf{x}) = \boldsymbol{\beta}^0 f(\mathbf{x}), \mathbf{p}(\mathbf{x}) = \mathbf{p}^0 f(\mathbf{x}), \boldsymbol{\tau}(\mathbf{x}) = \boldsymbol{\tau}^0 f(\mathbf{x}) \quad (3)$$

$$\boldsymbol{\chi}(\mathbf{x}) = \boldsymbol{\chi}^0 f(\mathbf{x}), \boldsymbol{\xi}(\mathbf{x}) = \boldsymbol{\xi}^0 f(\mathbf{x}), \boldsymbol{\gamma}(\mathbf{x}) = \boldsymbol{\gamma}^0 f(\mathbf{x}) \quad (4)$$

where  $\mathbf{C}$ ,  $\mathbf{e}$  and  $\mathbf{q}$  represent elastic, piezoelectric and piezomagnetic coefficient matrices, respectively.  $\boldsymbol{\varepsilon}$ ,  $\boldsymbol{\mu}$  and  $\mathbf{m}$  represent the dielectric, magnetic permeability and magneto-electric coefficient matrices, respectively.  $\boldsymbol{\beta}$ ,  $\boldsymbol{\chi}$ ,  $\mathbf{p}$ ,  $\boldsymbol{\xi}$ ,  $\boldsymbol{\tau}$  and  $\boldsymbol{\gamma}$  are the vectors of thermal expansion, moisture expansion, pyroelectric, hygroelectric, pyromagnetic and hygromagnetic coefficient, respectively.  $z$  denotes the value of space coordinates in  $z$ -direction.  $\mathbf{C}^0$ ,  $\mathbf{e}^0$ ,  $\mathbf{q}^0$ ,  $\boldsymbol{\varepsilon}^0$ ,  $\boldsymbol{\mu}^0$ ,  $\mathbf{m}^0$ ,  $\boldsymbol{\beta}^0$ ,  $\mathbf{p}^0$ ,  $\boldsymbol{\tau}^0$ ,  $\boldsymbol{\chi}^0$ ,  $\boldsymbol{\xi}^0$  and  $\boldsymbol{\gamma}^0$  are the corresponding material constants for the bottom margin of the structural geometry.  $f(\mathbf{x})$  denotes the exponential function, whose expression is

$$f(\mathbf{x}) = e^{\frac{\mathbf{x}}{w}} \quad (5)$$

where  $n$  and  $w$  denote the exponential factor and the thickness of the structure, respectively.  $z$  is the spatial coordinate.

For FGME structures, the constitutive equations are

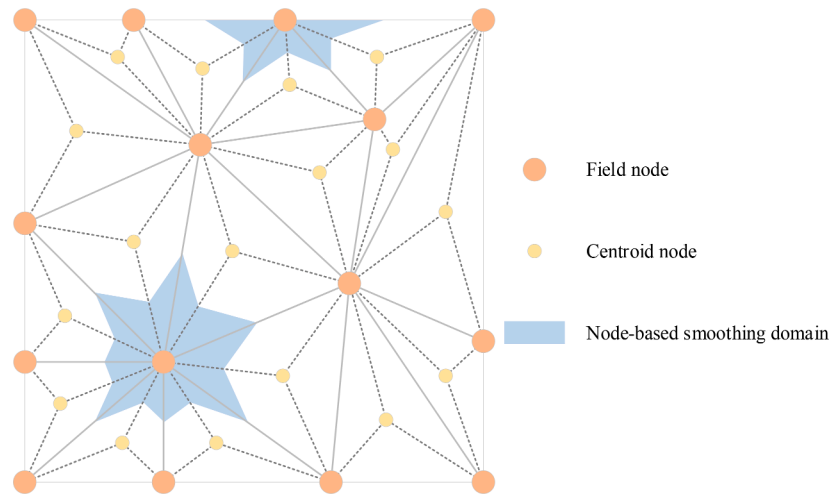


Fig. 2. Node-based smoothing domain.

$$\begin{Bmatrix} \boldsymbol{\sigma}(\mathbf{x}) \\ \mathbf{D}(\mathbf{x}) \\ \mathbf{B}(\mathbf{x}) \end{Bmatrix} = \begin{bmatrix} \mathbf{C}(\mathbf{x}) & -\mathbf{e}(\mathbf{x}) & -\mathbf{q}(\mathbf{x}) \\ \mathbf{e}^T(\mathbf{x}) & \boldsymbol{\varepsilon}(\mathbf{x}) & \mathbf{m}(\mathbf{x}) \\ \mathbf{q}^T(\mathbf{x}) & \mathbf{m}^T(\mathbf{x}) & \boldsymbol{\mu}(\mathbf{x}) \end{bmatrix} \begin{Bmatrix} \mathbf{S}(\mathbf{x}) \\ \mathbf{E}(\mathbf{x}) \\ \mathbf{H}(\mathbf{x}) \end{Bmatrix} + \begin{bmatrix} -\mathbf{C}(\mathbf{x})(\boldsymbol{\beta}(\mathbf{x})\Delta T + \boldsymbol{\chi}(\mathbf{x})\Delta m) \\ \mathbf{p}(\mathbf{x})\Delta T + \boldsymbol{\xi}(\mathbf{x})\Delta m \\ \boldsymbol{\tau}(\mathbf{x})\Delta T + \boldsymbol{\gamma}(\mathbf{x})\Delta m \end{bmatrix} \quad (6)$$

where  $\boldsymbol{\sigma}(\mathbf{x})$ ,  $\mathbf{D}(\mathbf{x})$  and  $\mathbf{B}(\mathbf{x})$  are the vectors of stress, electrical displacement and magnetic field intensity, respectively.  $\Delta T = T - T_0$ ,  $\Delta m = m - m_0$ ,  $T$  and  $T_0$  represent the final temperature and reference temperature,  $m$  and  $m_0$  represent the final moisture concentration and reference moisture concentration. The temperature and moisture concentration rise uniformly.

The equilibrium equations are

$$\nabla \cdot \boldsymbol{\sigma}(\mathbf{x}) = 0 \quad (7)$$

$$\nabla \cdot \mathbf{D}(\mathbf{x}) = 0 \quad (8)$$

$$\nabla \cdot \mathbf{B}(\mathbf{x}) = 0 \quad (9)$$

The geometric equations have the following formulations [57]

$$\mathbf{S}(\mathbf{x}) = \frac{1}{2}(\nabla \mathbf{u}(\mathbf{x}) + (\nabla \mathbf{u}(\mathbf{x}))^T) \quad (10)$$

$$\mathbf{E}(\mathbf{x}) = -\nabla \Phi(\mathbf{x}) \quad (11)$$

$$\mathbf{H}(\mathbf{x}) = -\nabla \Psi(\mathbf{x}) \quad (12)$$

Where  $\nabla$  is the Nabla operator.  $\mathbf{S}(\mathbf{x})$ ,  $\mathbf{u}(\mathbf{x})$  are the vector of strain and displacement,  $\mathbf{E}(\mathbf{x})$ ,  $\Phi(\mathbf{x})$  are the vector of electrical field and electric potential.  $\mathbf{H}(\mathbf{x})$ ,  $\Psi(\mathbf{x})$  are the vectors of magnetic flux and magnetic potential.

### 3. NS-RPIM

#### 3.1. Generalized displacement approximation using RPIM

Since the three-node triangular element is easy to be generated and has good adaptability to complex structures, it is used to discretize models in this paper [58]. To construct RPIM shape function, selecting local support nodes from different layers for interpolation is used by cell-based T2L-shcme[39].

The approximation  $w(\mathbf{x})$  of generalized displacement for FGME material is [59]

$$w(\mathbf{x}) = \sum_{i=1}^{N_s} R_i(\mathbf{x})a_i + \sum_{j=1}^3 p_j(\mathbf{x})b_j \quad (13)$$

where  $R_i(\mathbf{x})$ ,  $p_j(\mathbf{x})$  denotes multi-quadrics radial basis function (MQ-RBF) and PBF [60,61],  $N_s$  is the sum of two-layer local support nodes.  $a_i$  and  $b_j$  are the coefficients related to RBF and PBF but not known, respectively.

Owing to forcing each local support node to satisfy Eq. (13), we obtain

$$\mathbf{w} = \mathbf{R}_m \mathbf{a} + \mathbf{P}_m \mathbf{b} \quad (14)$$

where  $\mathbf{R}_m$  and  $\mathbf{P}_m$  are the moment matrices.

The constrain conditions are used to guarantee the unique solution to the coefficients

$$\mathbf{P}_m^T \mathbf{a} = 0 \quad (15)$$

Combining Eqs. (14) and (15)

$$\begin{bmatrix} \mathbf{R}_m & \mathbf{P}_m \\ \mathbf{P}_m^T & 0 \end{bmatrix} \begin{Bmatrix} \mathbf{a} \\ \mathbf{b} \end{Bmatrix} = \begin{Bmatrix} \mathbf{w} \\ 0 \end{Bmatrix} \quad (16)$$

Based on Eqs. (13) and (16)

$$w(\mathbf{x}) = \bar{\mathbf{N}}(\mathbf{x}) \begin{Bmatrix} \mathbf{w} \\ 0 \end{Bmatrix} \quad (17)$$

where

$$\bar{\mathbf{N}}(\mathbf{x}) = [\mathbf{R}^T(\mathbf{x}) \quad \mathbf{P}^T(\mathbf{x})] \begin{bmatrix} \mathbf{R}_m & \mathbf{P}_m \\ \mathbf{P}_m^T & 0 \end{bmatrix}^{-1} = [N_1(\mathbf{x}) \quad N_2(\mathbf{x}) \quad \cdots \quad N_{N_s+3}(\mathbf{x})] \quad (18)$$

The RPIM shape function is  $\mathbf{N}(\mathbf{x}) = [N_1(\mathbf{x}) \quad N_2(\mathbf{x}) \quad \cdots \quad N_{N_s}(\mathbf{x})]$ .

#### 3.2. Inhomogeneous node-based GST

Due to the limitation of low accuracy in dealing with multi-physics coupling problems using FEM, a new algorithm be developed to accurately analyze the responses of FGME structures. By linking the centroid nodes and the mid-edge points around node  $x_b$ , the node-based smoothing domain  $\Omega_i$  is created (Fig. 2). Using RPIM shape functions of displacement  $\mathbf{N}_i^u(\mathbf{x})$ , electric potential  $\mathbf{N}_i^\Phi(\mathbf{x})$  and magnetic potential  $\mathbf{N}_i^\Psi(\mathbf{x})$ . the displacement function  $\bar{\mathbf{u}}_i(\mathbf{x})$ , electric potential function  $\bar{\Phi}_i(\mathbf{x})$  and magnetic potential function  $\bar{\Psi}_i(\mathbf{x})$  are given as

$$\bar{\mathbf{u}}_i(\mathbf{x}) = \mathbf{N}_i^u(\mathbf{x})\mathbf{u}_i(\mathbf{x}) \quad (19)$$

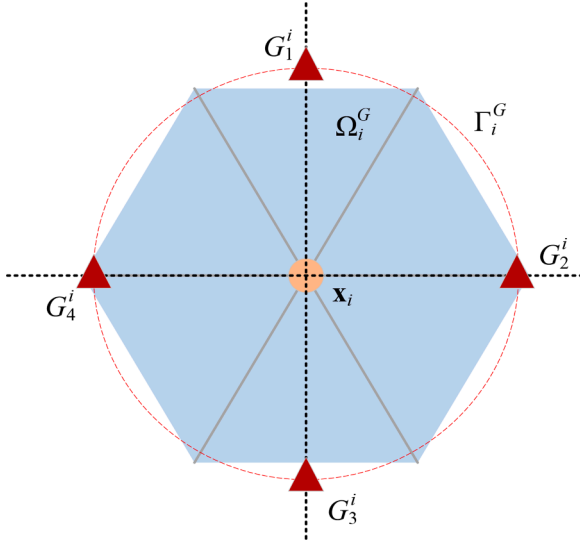


Fig. 3. The approximate domain of ISNS-RPIM.

Table 1

Material parameters of BaTiO<sub>3</sub>-CoFe<sub>2</sub>O<sub>4</sub> [25,63].

Elastic coefficient C-N/m <sup>2</sup>					
C <sub>11</sub>	C <sub>12</sub>	C <sub>13</sub>	C <sub>33</sub>	C <sub>44</sub>	C <sub>66</sub>
$1.66 \times 10^{11}$	$0.77 \times 10^{11}$	$0.78 \times 10^{11}$	$1.62 \times 10^{11}$	$0.43 \times 10^{11}$	$0.445 \times 10^{11}$
Piezoelectric coefficient e-C/m <sup>2</sup>			Piezomagnetic coefficient q-N/Am		
e <sub>31</sub>	e <sub>33</sub>	e <sub>15</sub>	q <sub>31</sub>	q <sub>33</sub>	q <sub>15</sub>
-4.4	18.6	11.6	580.3	699.7	550.0
Dielectric coefficient $\epsilon$ -C <sup>2</sup> /Nm <sup>2</sup>		Magneto-electric coefficient m-Ns/Vc		Magnetic permeability $\mu$ -Ns <sup>2</sup> /C <sup>2</sup>	
$\epsilon_{11}$	$\epsilon_{33}$	m <sub>11</sub>	m <sub>33</sub>	$\mu_{11}$	$\mu_{33}$
$1.12 \times 10^{-8}$	$1.26 \times 10^{-8}$	$5.0 \times 10^{-12}$	$3.0 \times 10^{-12}$	$5.0 \times 10^{-6}$	$10.0 \times 10^{-6}$
Pyroelectric coefficient p-C/m <sup>2</sup> K		Pyromagnetic coefficient $\tau$ -C/m <sup>2</sup> K		hygroelectric coefficient $\xi$ -Cm/kg	
p <sub>3</sub>	$\tau_3$	$\xi$	$\gamma$	hygromagnetic coefficient $\gamma$ -Nm <sup>2</sup> /Akg	
$-13 \times 10^{-5}$	$6 \times 10^{-3}$	0	0		
Thermal expansion coefficient $\beta$ -K <sup>-1</sup>		Moisture expansion coefficient $\chi$ -m <sup>3</sup> /kg		Density $\rho$ -kg/m <sup>3</sup>	
$\beta_1 = \beta_2$	$\beta_3$	$\chi_2 = \chi_3$	$\rho$		
$14.1 \times 10^{-6}$	$7.2 \times 10^{-6}$	$1.1 \times 10^{-4}$	5730		

$$\bar{\Phi}_i(\mathbf{x}) = \mathbf{N}_i^{\Phi}(\mathbf{x})\Phi_i(\mathbf{x}) \quad (20)$$

$$\bar{\Psi}_i(\mathbf{x}) = \mathbf{N}_i^{\Psi}(\mathbf{x})\Psi_i(\mathbf{x}) \quad (21)$$

The smoothed strain  $\bar{\mathbf{S}}_i(\mathbf{x})$ , smoothed electrical field intensity  $\bar{\mathbf{E}}_i(\mathbf{x})$  and smoothed magnetic flux density  $\bar{\mathbf{H}}_i(\mathbf{x})$  are given as

$$\bar{\mathbf{S}}_i(\mathbf{x}) = \frac{1}{A_i} \int_{\Gamma_i} \mathbf{L}_n^u \bar{\mathbf{u}}_i(\mathbf{x}) d\Gamma = \sum_{j=1}^{N_s} [\bar{\mathbf{B}}_i^u(\mathbf{x})]_j [\mathbf{u}_i(\mathbf{x})]_j \quad (22)$$

$$\bar{\mathbf{E}}_i(\mathbf{x}) = \frac{1}{A_i} \int_{\Gamma_i} \mathbf{L}_n^{\Phi} \bar{\Phi}_i(\mathbf{x}) d\Gamma = - \sum_{j=1}^{N_s} [\bar{\mathbf{B}}_i^{\Phi}(\mathbf{x})]_j [\Phi_i(\mathbf{x})]_j \quad (23)$$

$$\bar{\mathbf{H}}_i(\mathbf{x}) = \frac{1}{A_i} \int_{\Gamma_i} \mathbf{L}_n^{\Psi} \bar{\Psi}_i(\mathbf{x}) d\Gamma = - \sum_{j=1}^{N_s} [\bar{\mathbf{B}}_i^{\Psi}(\mathbf{x})]_j [\Psi_i(\mathbf{x})]_j \quad (24)$$

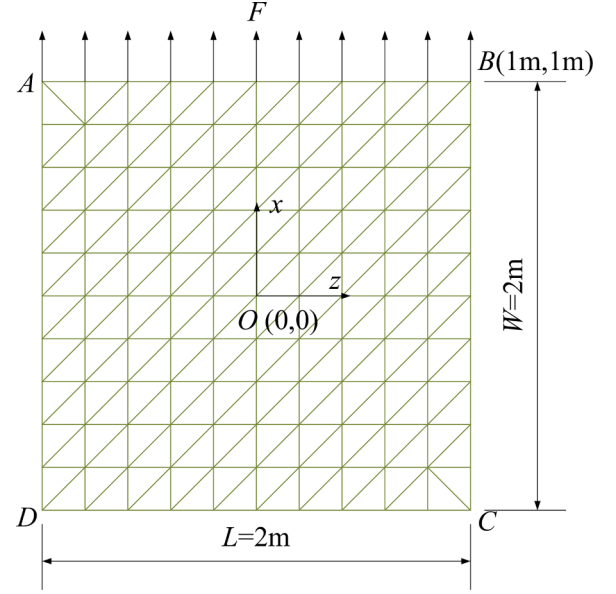


Fig. 4. Meshes and dimensions of the MEE plate.

where  $A_i$  denotes the area of  $\Omega_i$  and  $\Gamma_i$  denotes the boundary of  $\Omega_i$ .  $\mathbf{L}_n^u$ ,  $\mathbf{L}_n^{\Phi}$  and  $\mathbf{L}_n^{\Psi}$  are the unit normal vector matrices

$$\mathbf{L}_n^u = \begin{bmatrix} n_x & 0 & n_z \\ 0 & n_z & n_x \end{bmatrix}^T, \mathbf{L}_n^{\Phi} = \mathbf{L}_n^{\Psi} = \begin{bmatrix} n_x & n_z \end{bmatrix}^T \quad (25)$$

$[\bar{\mathbf{B}}_i^u(\mathbf{x})]_j$ ,  $[\bar{\mathbf{B}}_i^{\Phi}(\mathbf{x})]_j$  and  $[\bar{\mathbf{B}}_i^{\Psi}(\mathbf{x})]_j$  could be expressed as

$$[\bar{\mathbf{B}}_i^u(\mathbf{x})]_j = \begin{bmatrix} (\bar{b}_{jx}^u(\mathbf{x}))_i & 0 & (\bar{b}_{jz}^u(\mathbf{x}))_i \\ 0 & (\bar{b}_{jz}^u(\mathbf{x}))_i & (\bar{b}_{jx}^u(\mathbf{x}))_i \end{bmatrix}^T \quad (26)$$

$$[\bar{\mathbf{B}}_i^{\Phi}(\mathbf{x})]_j = [(\bar{b}_{jx}^{\Phi}(\mathbf{x}))_i \quad (\bar{b}_{jz}^{\Phi}(\mathbf{x}))_i]^T, [\bar{\mathbf{B}}_i^{\Psi}(\mathbf{x})]_j = [(\bar{b}_{jx}^{\Psi}(\mathbf{x}))_i \quad (\bar{b}_{jz}^{\Psi}(\mathbf{x}))_i]^T \quad (27)$$

Substituting  $\bar{\mathbf{S}}_i(\mathbf{x})$ ,  $\bar{\mathbf{E}}_i(\mathbf{x})$  and  $\bar{\mathbf{H}}_i(\mathbf{x})$  defined in Eqs. (7)–(9) by Eqs. (22)–(24), the stiffness matrices for FGME structures with  $N$  nodes can be computed as

$$\mathbf{K}_{uu} = \sum_{i=1}^N A_i [\bar{\mathbf{B}}_i^u(\mathbf{x})]^T \mathbf{C}(\mathbf{x}) [\bar{\mathbf{B}}_i^u(\mathbf{x})], \mathbf{K}_{\Phi\Phi} = \sum_{i=1}^N A_i [\bar{\mathbf{B}}_i^{\Phi}(\mathbf{x})]^T \boldsymbol{\epsilon}(\mathbf{x}) [\bar{\mathbf{B}}_i^{\Phi}(\mathbf{x})] \quad (28)$$

$$\mathbf{K}_{\Psi\Psi} = \sum_{i=1}^N A_i [\bar{\mathbf{B}}_i^{\Psi}(\mathbf{x})]^T \boldsymbol{\mu}(\mathbf{x}) [\bar{\mathbf{B}}_i^{\Psi}(\mathbf{x})], \mathbf{K}_{u\Phi} = \sum_{i=1}^N A_i [\bar{\mathbf{B}}_i^u(\mathbf{x})]^T \mathbf{e}(\mathbf{x}) [\bar{\mathbf{B}}_i^{\Phi}(\mathbf{x})] \quad (29)$$

$$\mathbf{K}_{u\Psi} = \sum_{i=1}^N A_i [\bar{\mathbf{B}}_i^u(\mathbf{x})]^T \mathbf{q}(\mathbf{x}) [\bar{\mathbf{B}}_i^{\Psi}(\mathbf{x})], \mathbf{K}_{\Phi\Psi} = \sum_{i=1}^N A_i [\bar{\mathbf{B}}_i^{\Phi}(\mathbf{x})]^T \mathbf{m}(\mathbf{x}) [\bar{\mathbf{B}}_i^{\Psi}(\mathbf{x})] \quad (30)$$

### 3.3. ISNS-RPIM

To guarantee stability and accuracy in solving dynamic problems, stable terms need to be introduced into NS-RPIM [42,62]. The stable terms are constructed based on  $\Omega_i^G$  (Fig. 3).  $\Omega_i^G$  defined as approximate domain is closed to a circular region  $\Omega_i^G$  whose area is the same as  $A_i$ . Then  $\Omega_i^G$  is subdivided to four sub-regions. The radius of the approximate domain is

$$l_c = \sqrt{A_i/\pi} \quad (31)$$

**Table 2**

Generalized displacement of point B.

	$u_x (10^{-10} \text{ m})$	$u_z (10^{-9} \text{ m})$	$\Phi (\text{V})$	$\Psi (10^{-2} \text{ A})$
ISNS-RPIM	-6.333170612063460	1.1366757464314900	1.8990980642225400	4.27812431865876
Analytical Solution	-6.3332000	1.1366760	1.8991000	4.278120

**Table 3**

The efficiency of EFGM, FEM and ISNS-RPIM [64].

Method	CPU time/s	Energy error (%)	EC
EFGM	6.004	$6.609 \times 10^{-5}$	$2.52 \times 10^3$
FEM	1.746	$3.047 \times 10^{-4}$	$1.88 \times 10^3$
ISNS-RPIM	16.386	$7.223 \times 10^{-6}$	$8.45 \times 10^3$

The gradient of  $u$ ,  $\Phi$  and  $\Psi$  at node  $x_i$  are

$$\nabla \mathbf{u} = \nabla \mathbf{u}_i + \frac{\partial(\nabla \mathbf{u})}{\partial x}(x - x_i) + \frac{\partial(\nabla \mathbf{u})}{\partial z}(z - z_i) \quad (32)$$

$$\nabla \Phi = \nabla \Phi_i + \frac{\partial(\nabla \Phi)}{\partial x}(x - x_i) + \frac{\partial(\nabla \Phi)}{\partial z}(z - z_i) \quad (33)$$

$$\nabla \Psi = \nabla \Psi_i + \frac{\partial(\nabla \Psi)}{\partial x}(x - x_i) + \frac{\partial(\nabla \Psi)}{\partial z}(z - z_i) \quad (34)$$

The gradient items of complementary integration points  $G_k^i$  ( $k=1, 2, 3, 4$ ) are

$$(\nabla \mathbf{u})_{i1} = \nabla \mathbf{u}_i + l_c \cdot \partial(\nabla \mathbf{u}) / \partial x, (\nabla \mathbf{u})_{i2} = \nabla \mathbf{u}_i + l_c \cdot \partial(\nabla \mathbf{u}) / \partial z \quad (35)$$

$$(\nabla \mathbf{u})_{i3} = \nabla \mathbf{u}_i - l_c \cdot \partial(\nabla \mathbf{u}) / \partial x, (\nabla \mathbf{u})_{i4} = \nabla \mathbf{u}_i - l_c \cdot \partial(\nabla \mathbf{u}) / \partial z \quad (36)$$

$$(\nabla \Phi)_{i1} = \nabla \Phi_i + l_c \cdot \partial(\nabla \Phi) / \partial x, (\nabla \Phi)_{i2} = \nabla \Phi_i + l_c \cdot \partial(\nabla \Phi) / \partial z \quad (37)$$

$$(\nabla \Phi)_{i3} = \nabla \Phi_i - l_c \cdot \partial(\nabla \Phi) / \partial x, (\nabla \Phi)_{i4} = \nabla \Phi_i - l_c \cdot \partial(\nabla \Phi) / \partial z \quad (38)$$

$$(\nabla \Psi)_{i1} = \nabla \Psi_i + l_c \cdot \partial(\nabla \Psi) / \partial x, (\nabla \Psi)_{i2} = \nabla \Psi_i + l_c \cdot \partial(\nabla \Psi) / \partial z \quad (39)$$

$$(\nabla \Psi)_{i3} = \nabla \Psi_i - l_c \cdot \partial(\nabla \Psi) / \partial x, (\nabla \Psi)_{i4} = \nabla \Psi_i - l_c \cdot \partial(\nabla \Psi) / \partial z \quad (40)$$

The modified stiffness matrices are obtained by Eqs. (35)–(40) as follow

$$\bar{\mathbf{K}}_{uu} = \mathbf{K}_{uu} + \mathbf{K}_{uu}^*, \bar{\mathbf{K}}_{\Phi\Phi} = \mathbf{K}_{\Phi\Phi} + \mathbf{K}_{\Phi\Phi}^*, \bar{\mathbf{K}}_{\Psi\Psi} = \mathbf{K}_{\Psi\Psi} + \mathbf{K}_{\Psi\Psi}^*, \quad (41)$$

$$\bar{\mathbf{K}}_{u\Phi} = \mathbf{K}_{u\Phi} + \mathbf{K}_{u\Phi}^*, \bar{\mathbf{K}}_{u\Psi} = \mathbf{K}_{u\Psi} + \mathbf{K}_{u\Psi}^*, \bar{\mathbf{K}}_{\Phi\Psi} = \mathbf{K}_{\Phi\Psi} + \mathbf{K}_{\Phi\Psi}^*, \quad (42)$$

$$\mathbf{K}_{uu}^* = \sum_{i=1}^N \frac{A_i}{2} \left[ [\bar{\mathbf{B}}_i^u]^T \mathbf{C}(\mathbf{x}) [\bar{\mathbf{B}}_i^u]_x + [\bar{\mathbf{B}}_i^u]^T \mathbf{C}(\mathbf{x}) [\bar{\mathbf{B}}_i^u]_z \right] \quad (43)$$

$$\mathbf{K}_{u\Phi}^* = \sum_{i=1}^N \frac{A_i}{2} \left[ [\bar{\mathbf{B}}_i^u]^T \mathbf{e}(\mathbf{x}) [\bar{\mathbf{B}}_i^\Phi]_x + [\bar{\mathbf{B}}_i^u]^T \mathbf{e}(\mathbf{x}) [\bar{\mathbf{B}}_i^\Phi]_z \right] \quad (44)$$

$$\mathbf{K}_{u\Psi}^* = \sum_{i=1}^N \frac{A_i}{2} \left[ [\bar{\mathbf{B}}_i^u]^T \mathbf{q}(\mathbf{x}) [\bar{\mathbf{B}}_i^\Psi]_x + [\bar{\mathbf{B}}_i^u]^T \mathbf{q}(\mathbf{x}) [\bar{\mathbf{B}}_i^\Psi]_z \right] \quad (45)$$

$$\mathbf{K}_{\Phi\Phi}^* = \sum_{i=1}^N \frac{A_i}{2} \left[ [\bar{\mathbf{B}}_i^\Phi]^T \mathbf{e}(\mathbf{x}) [\bar{\mathbf{B}}_i^\Phi]_x + [\bar{\mathbf{B}}_i^\Phi]^T \mathbf{e}(\mathbf{x}) [\bar{\mathbf{B}}_i^\Phi]_z \right] \quad (46)$$

$$\mathbf{K}_{\Psi\Psi}^* = \sum_{i=1}^N \frac{A_i}{2} \left[ [\bar{\mathbf{B}}_i^\Psi]^T \mathbf{q}(\mathbf{x}) [\bar{\mathbf{B}}_i^\Psi]_x + [\bar{\mathbf{B}}_i^\Psi]^T \mathbf{q}(\mathbf{x}) [\bar{\mathbf{B}}_i^\Psi]_z \right] \quad (47)$$

$$\mathbf{K}_{\Phi\Psi}^* = \sum_{i=1}^N \frac{A_i}{2} \left[ [\bar{\mathbf{B}}_i^\Phi]^T \mathbf{m}(\mathbf{x}) [\bar{\mathbf{B}}_i^\Psi]_x + [\bar{\mathbf{B}}_i^\Phi]^T \mathbf{m}(\mathbf{x}) [\bar{\mathbf{B}}_i^\Psi]_z \right] \quad (48)$$

The discretized system equations are

$$[\bar{\mathbf{K}}_{uu}]\{\mathbf{u}\} + [\bar{\mathbf{K}}_{u\Phi}]\{\Phi\} + [\bar{\mathbf{K}}_{u\Psi}]\{\Psi\} = \{\mathbf{F}_u\} + \{\mathbf{F}_H\} + \{\mathbf{F}_T\} \quad (49)$$

$$[\bar{\mathbf{K}}_{\Phi u}]\{\mathbf{u}\} - [\bar{\mathbf{K}}_{\Phi\Phi}]\{\Phi\} - [\bar{\mathbf{K}}_{\Phi\Psi}]\{\Psi\} = -\{\mathbf{F}_{he}\} - \{\mathbf{F}_{pe}\} \quad (50)$$

$$[\bar{\mathbf{K}}_{\Psi u}]\{\mathbf{u}\} - [\bar{\mathbf{K}}_{\Psi\Phi}]\{\Phi\} - [\bar{\mathbf{K}}_{\Psi\Psi}]\{\Psi\} = -\{\mathbf{F}_{hm}\} - \{\mathbf{F}_{pm}\} \quad (51)$$

where  $\{\mathbf{F}_u\}$ ,  $\{\mathbf{F}_H\}$  and  $\{\mathbf{F}_T\}$  are the load vector of mechanical, hygroscopic and thermal, respectively.  $\{\mathbf{F}_{he}\}$ ,  $\{\mathbf{F}_{pe}\}$ ,  $\{\mathbf{F}_{hm}\}$  and  $\{\mathbf{F}_{pm}\}$  denote load vectors of hygroelectric, pyroelectric, hygromagnetic, pyromagnetic, respectively. The expressions are

$$\{\mathbf{F}_T\} = \sum_{i=1}^N A_i [\bar{\mathbf{B}}_i^u(\mathbf{x})]^T \mathbf{C}(\mathbf{x}) \beta(\mathbf{x}) \Delta T, \{\mathbf{F}_H\} = \sum_{i=1}^N A_i [\bar{\mathbf{B}}_i^u(\mathbf{x})]^T \mathbf{C}(\mathbf{x}) \chi(\mathbf{x}) \Delta m \quad (52)$$

$$\{\mathbf{F}_{pe}\} = \sum_{i=1}^N A_i [\bar{\mathbf{B}}_i^\Phi(\mathbf{x})]^T \mathbf{p}(\mathbf{x}) \Delta T, \{\mathbf{F}_{pm}\} = \sum_{i=1}^N A_i [\bar{\mathbf{B}}_i^\Psi(\mathbf{x})]^T \boldsymbol{\tau}(\mathbf{x}) \Delta T \quad (53)$$

$$\{\mathbf{F}_{he}\} = \sum_{i=1}^N A_i [\bar{\mathbf{B}}_i^\Phi(\mathbf{x})]^T \boldsymbol{\xi}(\mathbf{x}) \Delta m, \{\mathbf{F}_{hm}\} = \sum_{i=1}^N A_i [\bar{\mathbf{B}}_i^\Psi(\mathbf{x})]^T \boldsymbol{\gamma}(\mathbf{x}) \Delta m \quad (54)$$

For free vibration problems, the right sides of Eqs. (49) to (51) are zero, and Eq. (49) is rewritten as

$$[\mathbf{M}]\{\ddot{\mathbf{u}}\} + [\bar{\mathbf{K}}_{uu}]\{\mathbf{u}\} + [\bar{\mathbf{K}}_{u\Phi}]\{\Phi\} + [\bar{\mathbf{K}}_{u\Psi}]\{\Psi\} = 0 \quad (55)$$

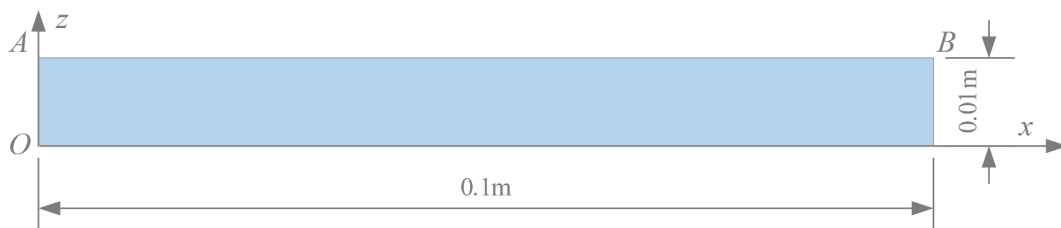
where  $\ddot{\mathbf{u}}$  is the acceleration vector,  $\mathbf{M}$  is the mass matrix. whose expressions are

$$\mathbf{M} = \sum_{N_e} [\text{diag}\{m_1 \quad m_1 \quad m_2 \quad m_2 \quad m_3 \quad m_3\}] \quad (56)$$

where  $N_e$  is the count of elements in the model.

Using Eqs. (49)–(51) and the condensation method [28], the equivalent stiffness matrix  $[\mathbf{K}_{eq}]$ , equivalent load vector  $\{\mathbf{F}_{eq}\}$  and mass matrix  $[\mathbf{M}]$  satisfy the following equation

$$[\mathbf{K}_{eq}]\{\mathbf{u}\} = \{\mathbf{F}_{eq}\} \quad (57)$$

**Fig. 5.** FGME clamped-clamped beam.

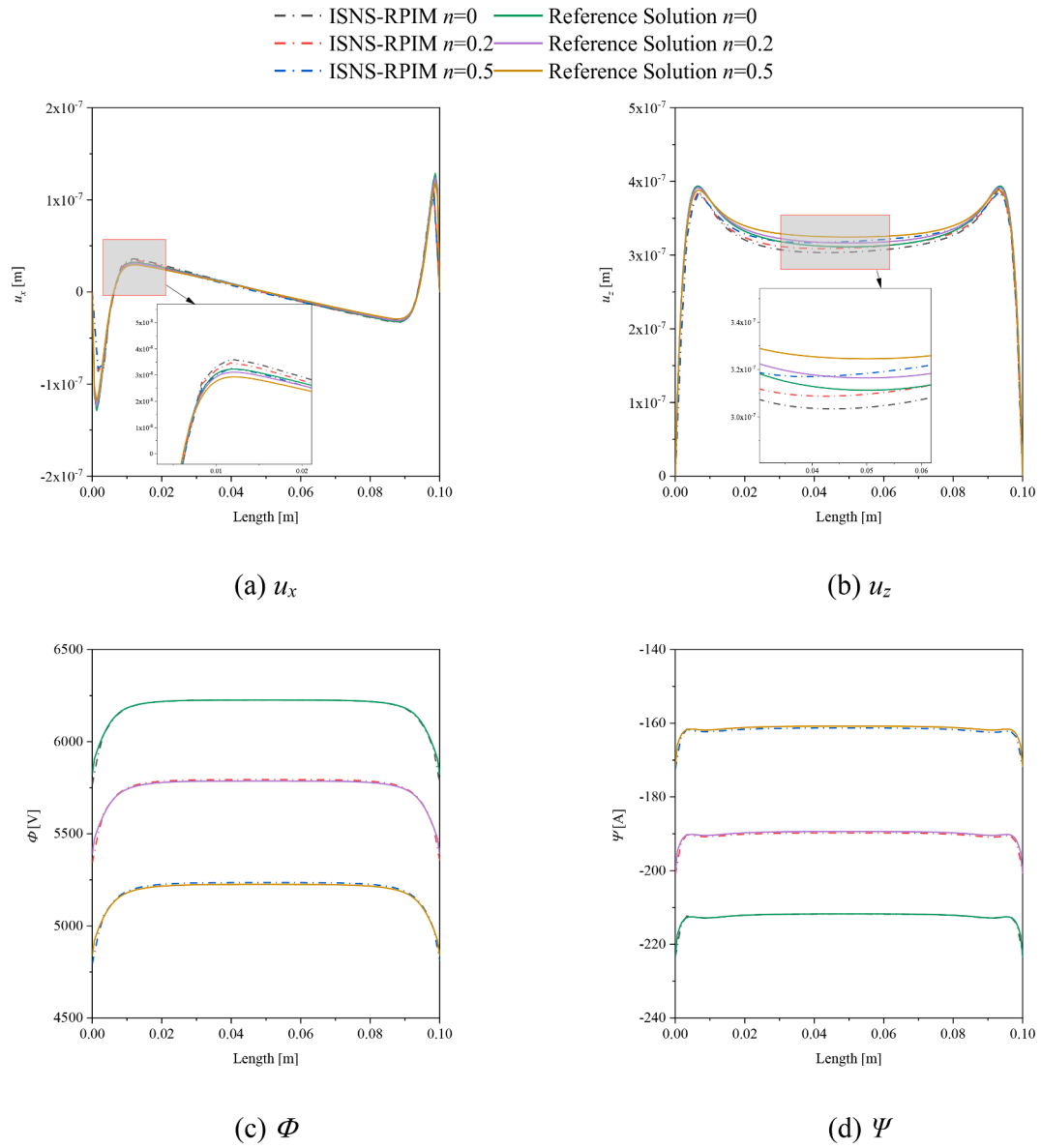


Fig. 6. The generalized displacement on edge AB.

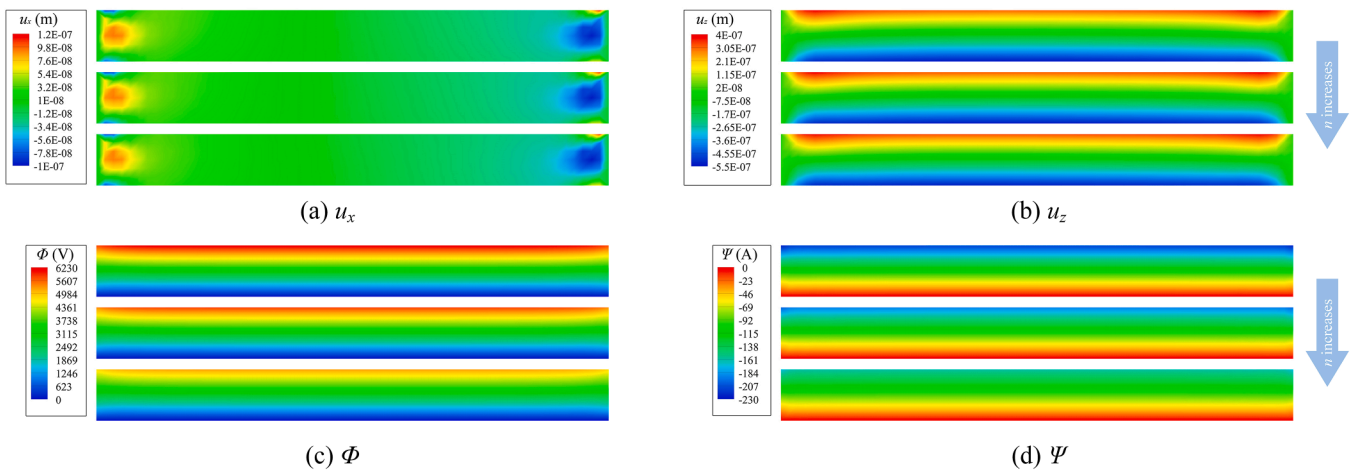


Fig. 7. FGME clamped-clamped beam contours.



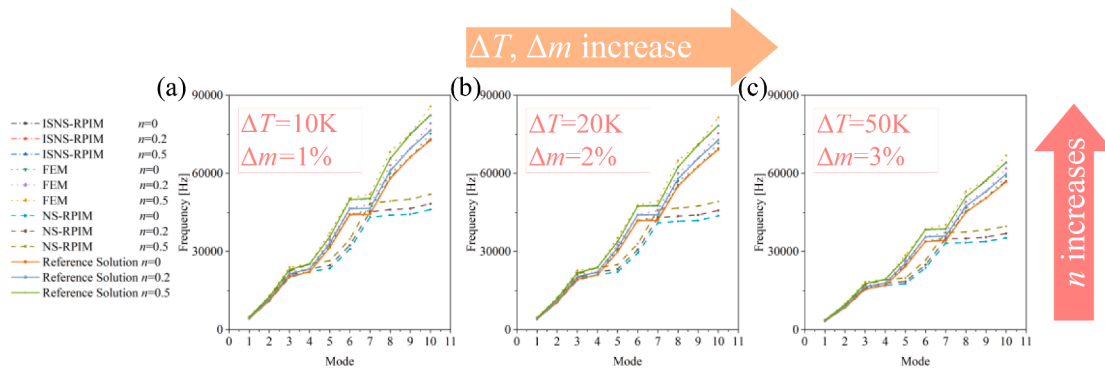


Fig. 8. Natural frequency of the FGME clamped-clamped beam with different hygrothermal loading. (a)  $\Delta T = 10$  K,  $\Delta m = 1\%$ , (b)  $\Delta T = 20$  K,  $\Delta m = 2\%$ , (c)  $\Delta T = 50$  K,  $\Delta m = 3\%$ .

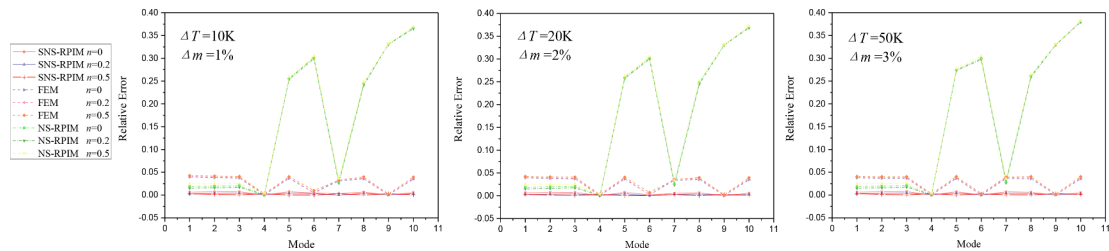


Fig. 9. Relative error of the FGME clamped-clamped beam with different  $n$ .

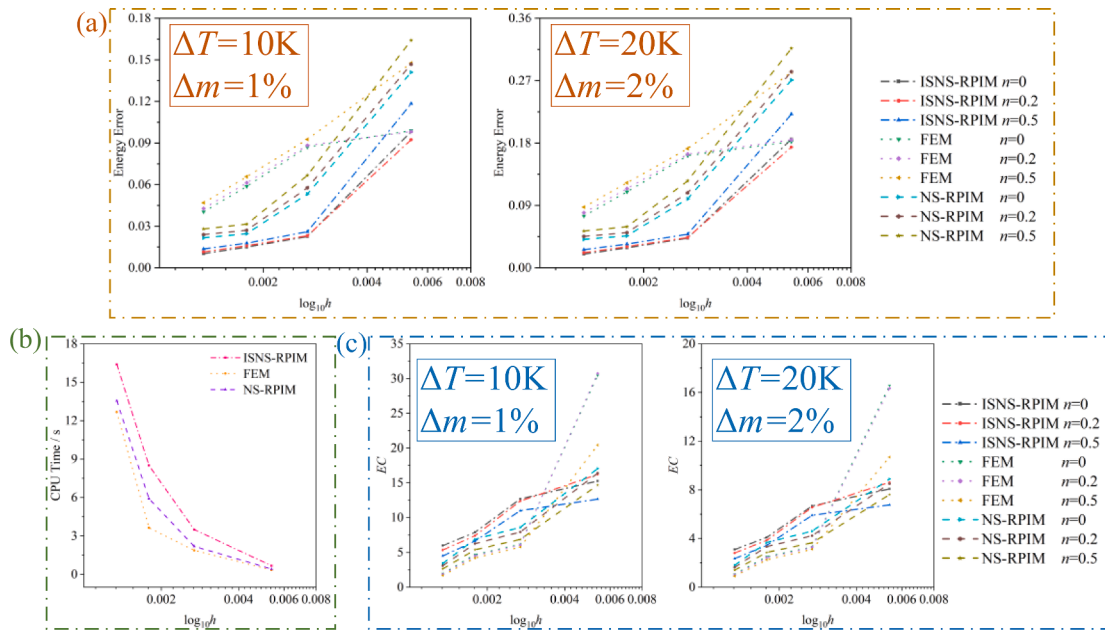


Fig. 10. (a) Energy error with different  $n$  ( $n=0, 0.2, 0.5$ ) (b) CPU times (c) Efficiency with different  $n$  ( $n=0, 0.2, 0.5$ ).

$$[\mathbf{K}_{eq}]\{\mathbf{u}\} + [\mathbf{M}]\{\ddot{\mathbf{u}}\} = 0, \text{ for free vibration problems} \quad (58)$$

The expression of  $\Phi$  and  $\Psi$  and the condensation technique are written as

$$\{\Phi\} = -[\mathbf{K}_{II}]^{-1}[\mathbf{F}_{\Phi_{sol}}] + [\mathbf{K}_{II}]^{-1}[\mathbf{K}_I]\{\mathbf{u}\} \quad (59)$$

$$[\mathbf{K}_I] = [\bar{\mathbf{K}}_{\Phi u}] - [\bar{\mathbf{K}}_{\Psi \Phi}][\bar{\mathbf{K}}_{\Psi \Psi}]^{-1}[\bar{\mathbf{K}}_{\Psi u}] \quad (60)$$

$$[\mathbf{K}_{II}] = [\bar{\mathbf{K}}_{\Phi \Phi}] - [\bar{\mathbf{K}}_{\Phi \Psi}][\bar{\mathbf{K}}_{\Psi \Psi}]^{-1}[\bar{\mathbf{K}}_{\Phi \Psi}] \quad (61)$$

$$\{\Psi\} = [\bar{\mathbf{K}}_{\Psi \Psi}]^{-1}[\bar{\mathbf{K}}_{u \Psi}]\{\mathbf{u}\} - [\bar{\mathbf{K}}_{\Psi \Psi}]^{-1}[\bar{\mathbf{K}}_{\Phi \Psi}]\{\Phi\} - [\bar{\mathbf{K}}_{\Psi \Psi}]^{-1}[\{\mathbf{F}_{pm}\} + \{\mathbf{F}_{hm}\}] \quad (62)$$

$$[\mathbf{K}_{III}] = [\mathbf{K}_{II}]^{-1}[\mathbf{K}_I] \quad (63)$$

$$[\mathbf{K}_{IV}] = [\mathbf{K}_{II}]^{-1}[\bar{\mathbf{K}}_{\Psi \Phi}][\bar{\mathbf{K}}_{\Psi \Psi}] \quad (64)$$

$$[\mathbf{K}_V] = [\bar{\mathbf{K}}_{uu}] + [\bar{\mathbf{K}}_{u \Psi}][\bar{\mathbf{K}}_{\Psi \Psi}]^{-1}[\bar{\mathbf{K}}_{\Psi u}] \quad (65)$$

$$[\mathbf{K}_{VI}] = [\bar{\mathbf{K}}_{u \Phi}] - [\bar{\mathbf{K}}_{u \Psi}][\bar{\mathbf{K}}_{\Psi \Psi}]^{-1}[\bar{\mathbf{K}}_{\Phi \Psi}] \quad (66)$$

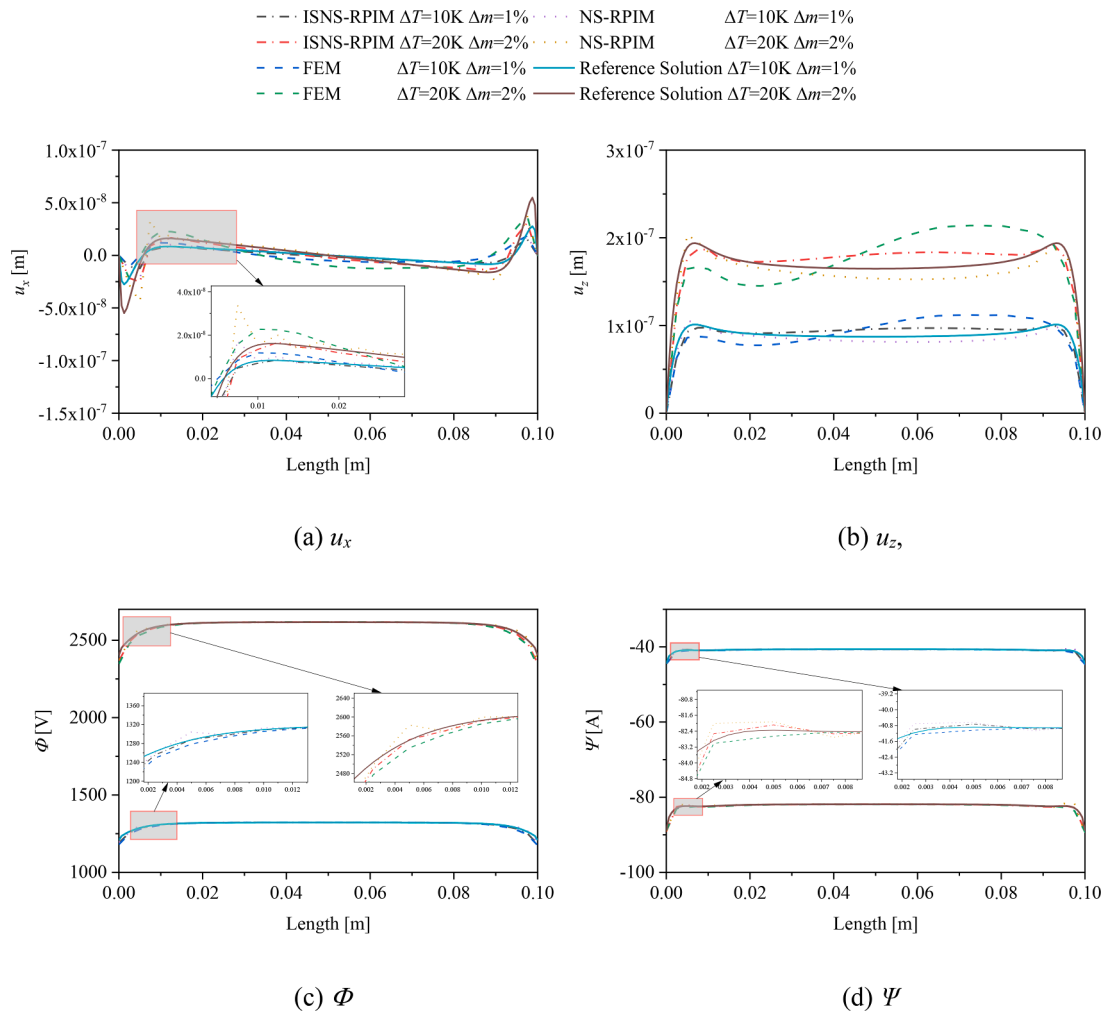


Fig. 11. The generalized displacement on edge AB (using distorted mesh).

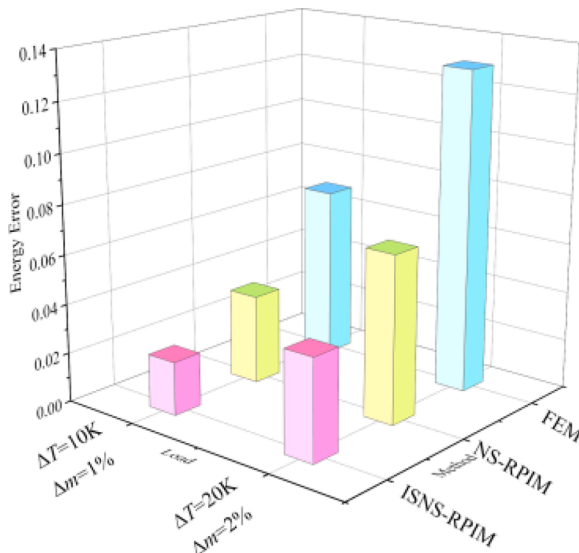


Fig. 12. The energy error using distorted mesh.

$$[\mathbf{K}_{VII}] = [\mathbf{K}_V] + [\mathbf{K}_{VI}][\mathbf{K}_{III}] \quad (67)$$

$$[\mathbf{K}_{VIII}] = [\mathbf{K}_{eq}] = [\mathbf{K}_{VI}][\mathbf{K}_{II}]^{-1} \quad (68)$$

$$[\mathbf{K}_{IX}] = [\bar{\mathbf{K}}_{u\psi}][\bar{\mathbf{K}}_{\psi\psi}]^{-1} - [\bar{\mathbf{K}}_{VI}][\bar{\mathbf{K}}_{IV}] \quad (69)$$

$$[\mathbf{F}_{eq}] = [\mathbf{K}_{IX}][\{\mathbf{F}_{pm}\} + \{\mathbf{F}_{hm}\}] + [\mathbf{K}_{VIII}][\{\mathbf{F}_{pe}\} + \{\mathbf{F}_{he}\}] + \{\mathbf{F}_u\} + \{\mathbf{F}_T\} + \{\mathbf{F}_H\} \quad (70)$$

$$\{\mathbf{F}_{\Phi_{sol}}\} = [\{\mathbf{F}_{pe}\} + \{\mathbf{F}_{he}\}] - [\bar{\mathbf{K}}_{\psi\Phi}]^T[\bar{\mathbf{K}}_{\psi\psi}]^{-1}[\{\mathbf{F}_{pm}\} + \{\mathbf{F}_{hm}\}] \quad (71)$$

#### 4. Numerical Examples

FGMEE structures subject to mechanical and hygrothermal loadings are studied using ISNS-RPIM in this section. Because of the small variation in temperature and moisture concentration, the following coefficients  $\beta$ ,  $p$ ,  $\tau$ ,  $\chi$ ,  $\xi$  and  $\gamma$  are neglected. First, the correctness of ISNS-RPIM is validated compared to the analytical solution. Then, several examples illustrate the advantages of the algorithm. Finally, the effects of exponential factor, hygrothermal loadings and empirical constants on the responses of FGMEE structures are analyzed.

The values of  $\xi$  and  $\gamma$  are zero. Table 1 gives the material constants of  $\text{BaTiO}_3\text{-CoFe}_2\text{O}_4$ . The influence of the empirical constants related to temperature and moisture dependency ( $\alpha^*$ ,  $\beta^*$ ) is considered. The



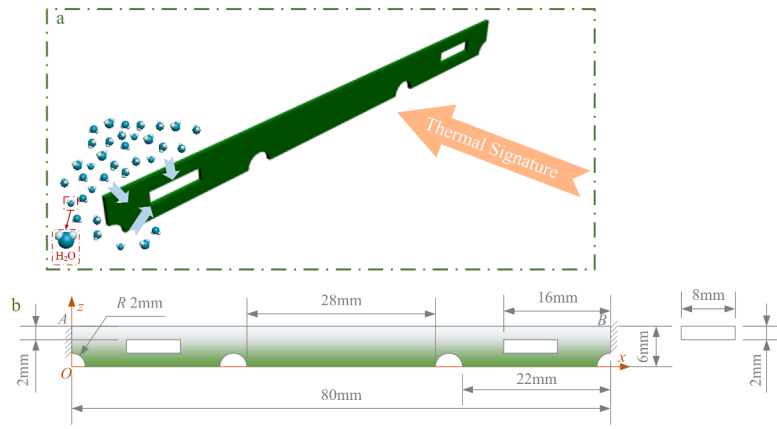


Fig. 13. (a) FGME energy harvester (b) Dimensions of the FGME energy harvester.

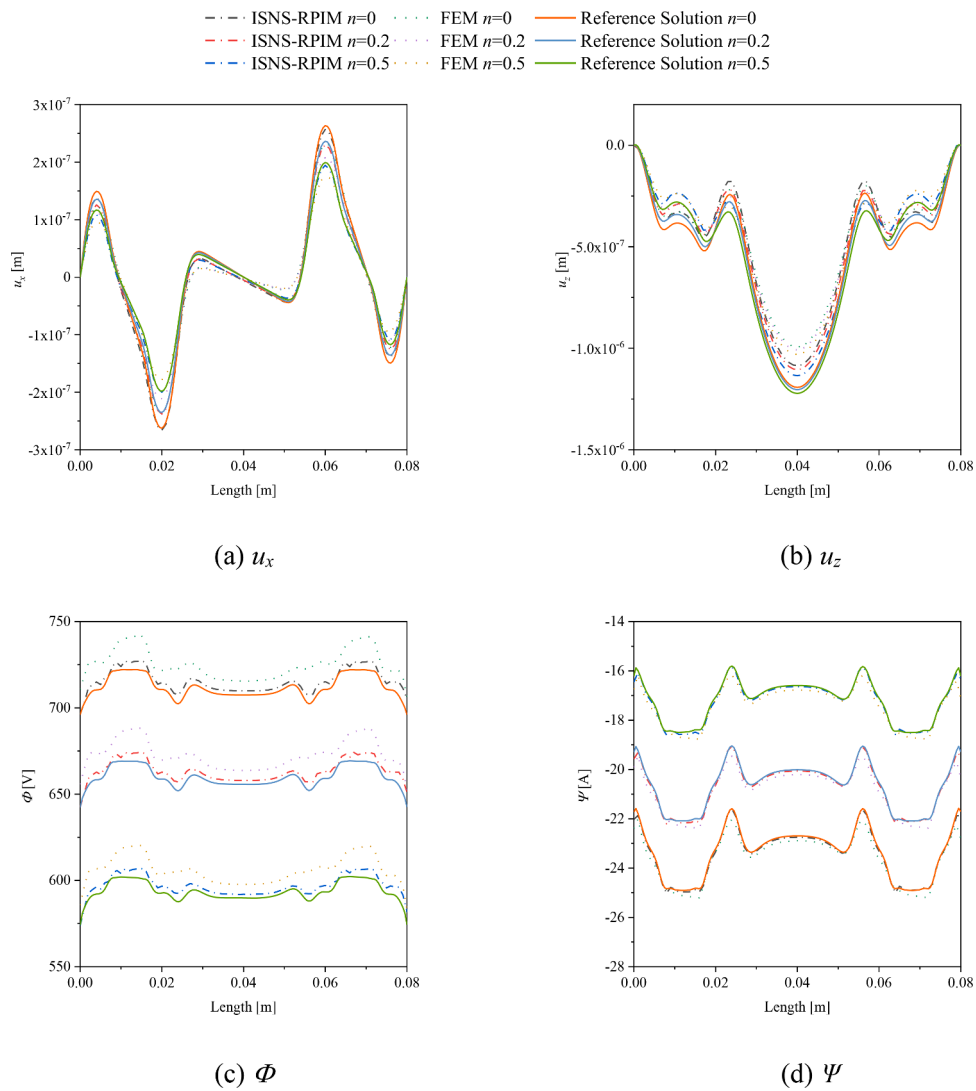


Fig. 14. The generalized displacement on edge AB ( $\Delta T = 10$  K,  $\Delta m = 1\%$ ).

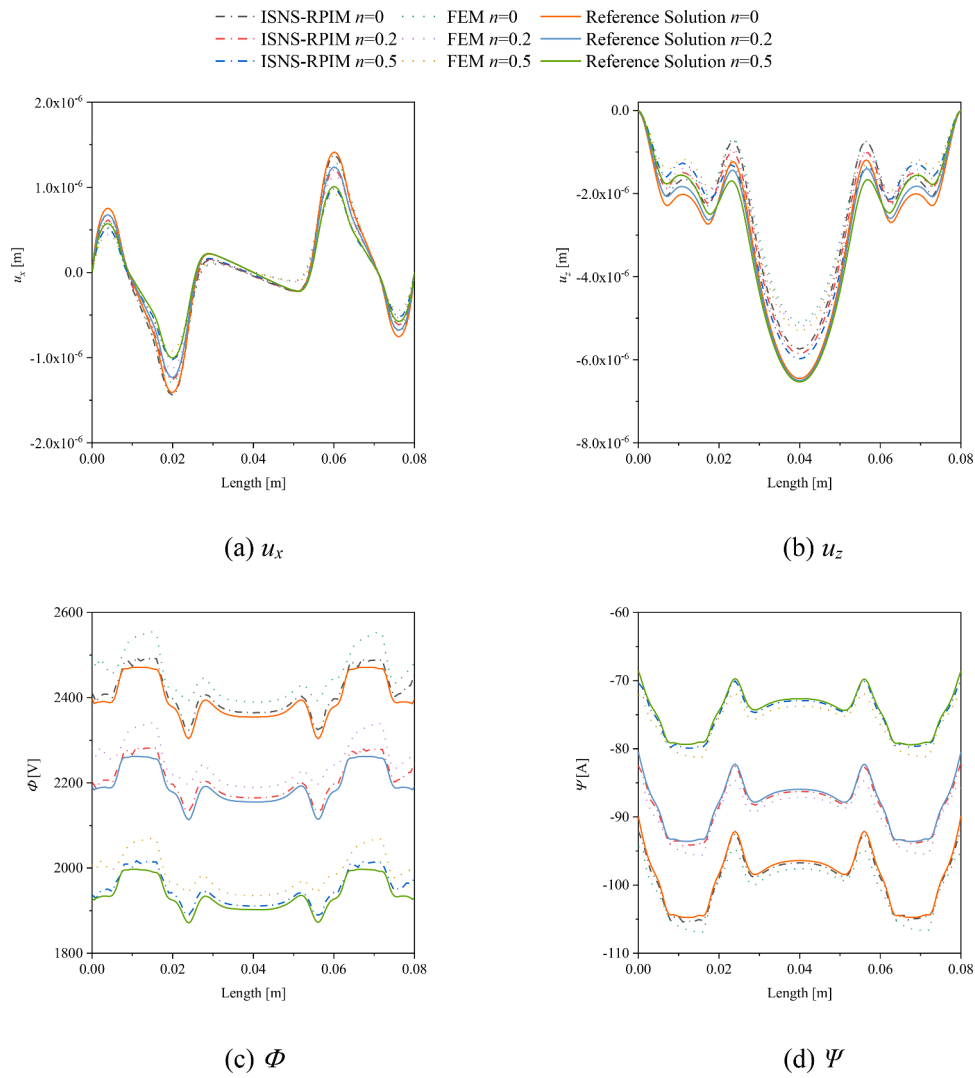


Fig. 15. The generalized displacement on edge AB ( $\Delta T = 40$  K,  $\Delta m = 3\%$ ).

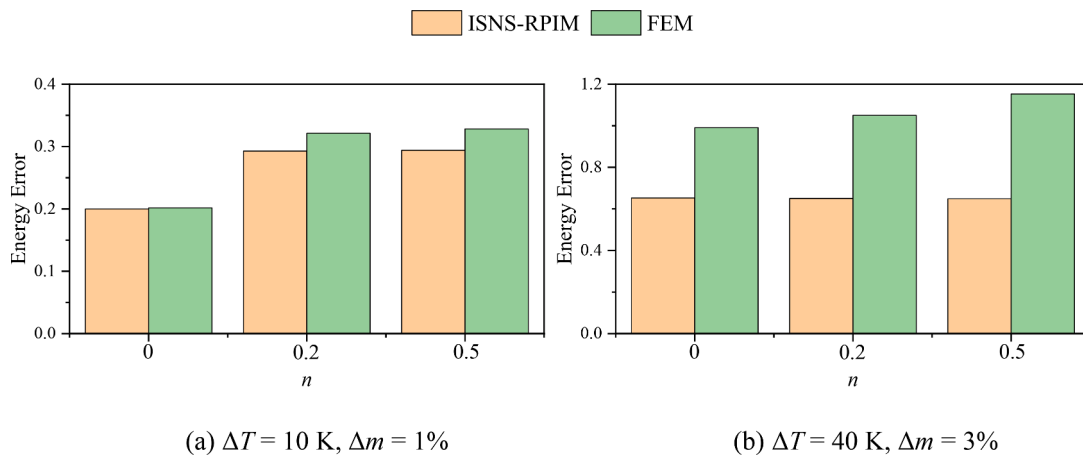


Fig. 16. The energy error of ISNS-RPIM and FEM.

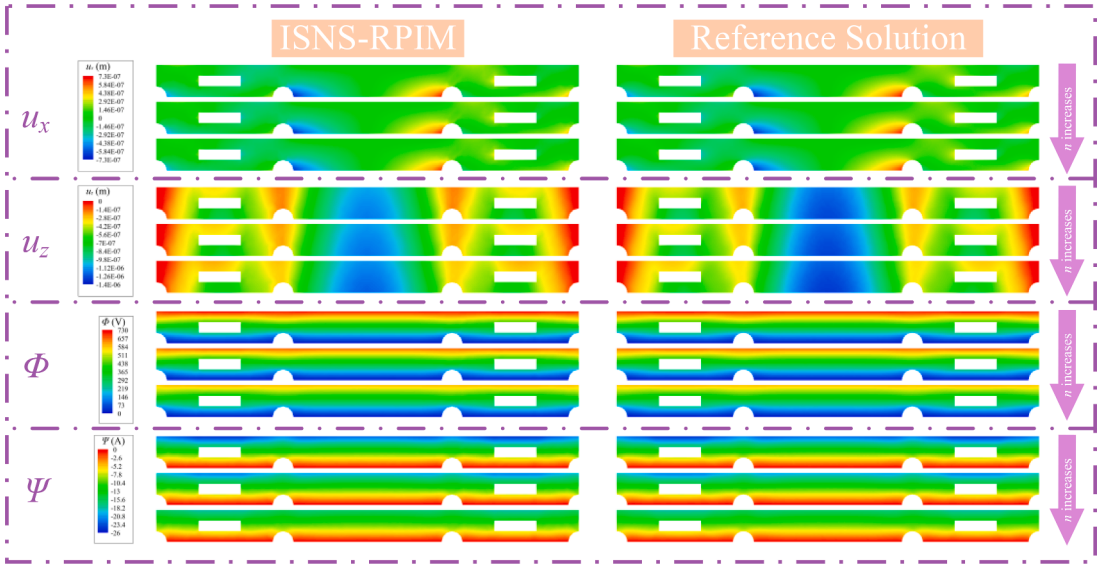


Fig. 17. Contours of the FGME sensor ( $\Delta T = 10$  K,  $\Delta m = 1\%$ ).

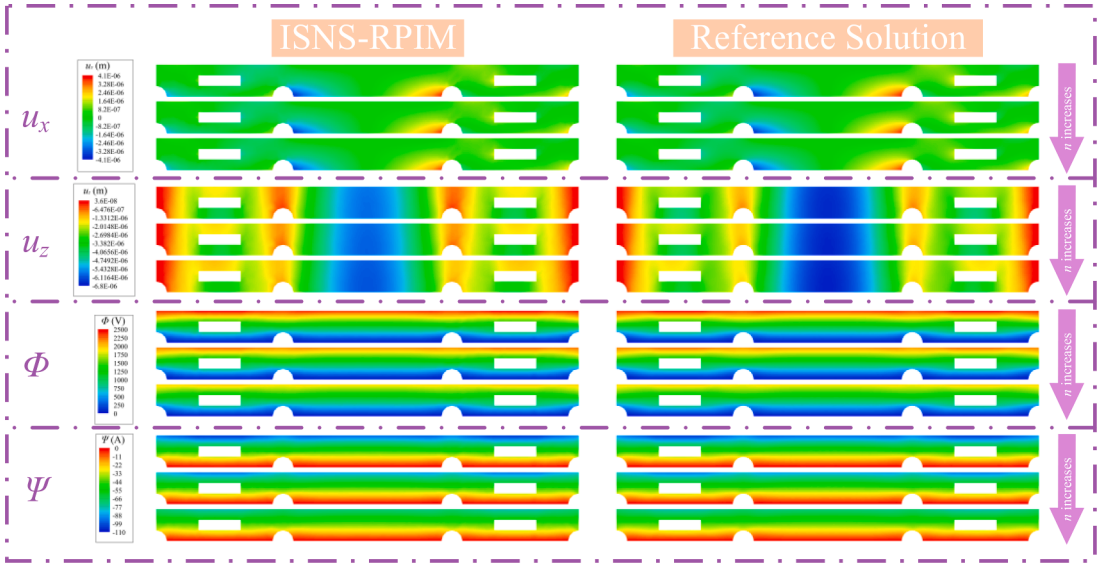


Fig. 18. Contours of the FGME sensor ( $\Delta T = 40$  K,  $\Delta m = 3\%$ ).

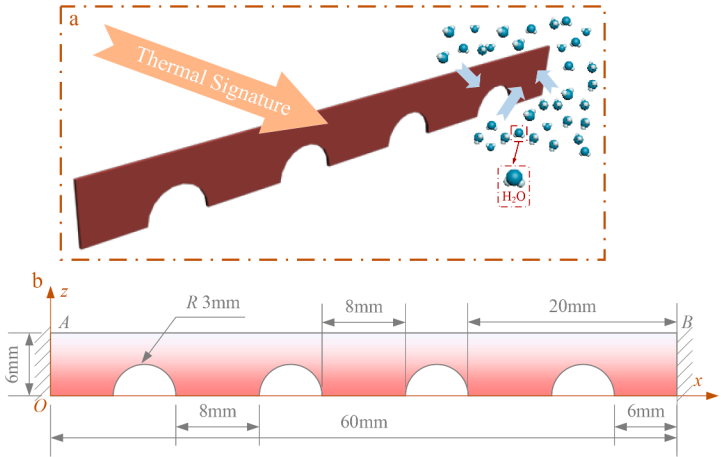


Fig. 19. (a) FGME energy harvester (b) Dimensions of the FGME energy harvester.

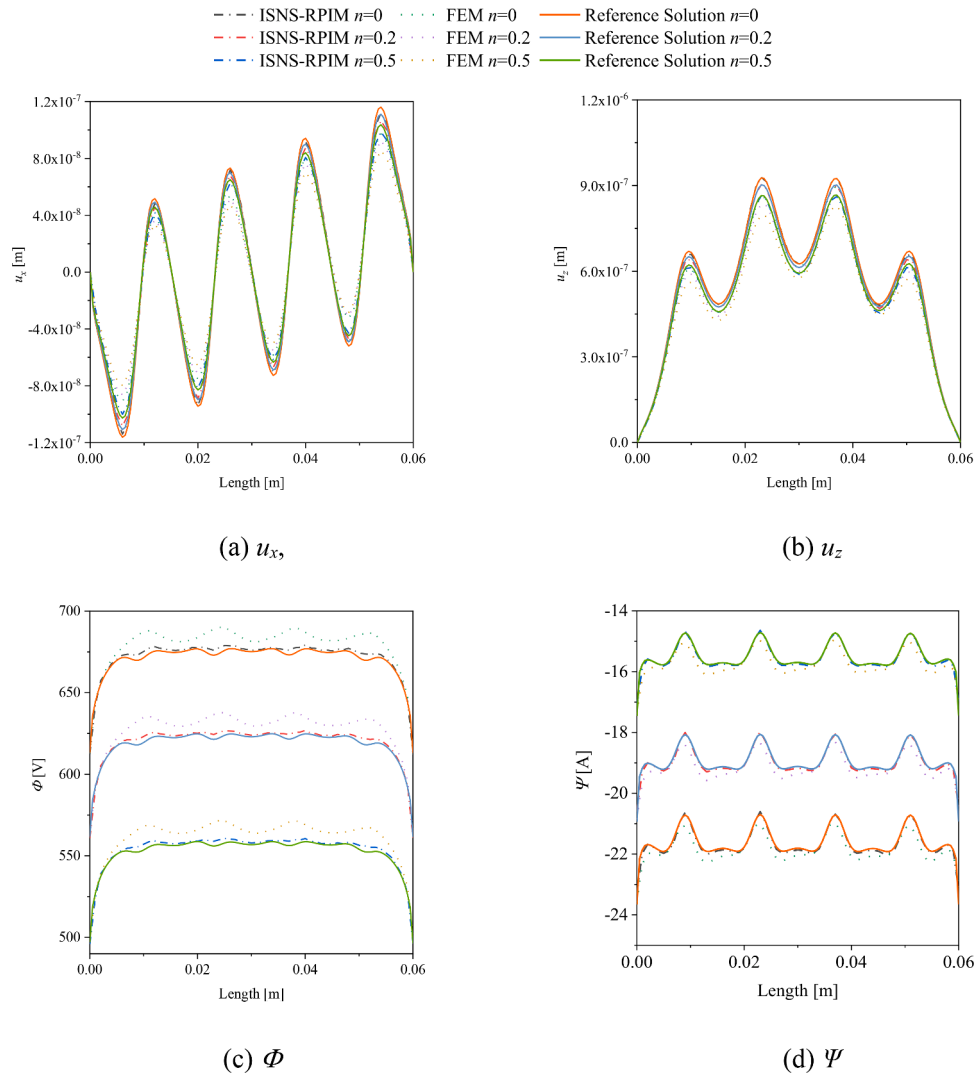


Fig. 20. The generalized displacement on edge AB ( $\alpha^* = \beta^* = 0$ ).

dependent elastic stiffness coefficient  $\tilde{C}$  and the independent elastic stiffness coefficient  $C^*$  have the following relationship [28]

$$\tilde{C} = C^* + C^* \alpha^* \Delta T + C^* \beta^* \Delta m \quad (72)$$

The error in total energy norm is introduced to verify the convergence and accuracy of ISNS-RPIM:

$$EC = \frac{1}{\text{CPUtime} \times \text{EnergyError}} \quad (75)$$

Then, create the irregular meshes ( $x_{ir}$ ,  $z_{ir}$ ) as follow [56].

$$x_{ir} = x + r_A \cdot \Delta x \cdot \alpha_{ir} \quad (76)$$

$$\text{EnergyError} = \sqrt{\frac{1}{2} \int \left( C_{ijkl} (S_{ij} - S_{ij}^{\text{An}}) (S_{kl} - S_{kl}^{\text{An}}) + \varepsilon_{ij} (E_i - E_i^{\text{An}}) (E_j - E_j^{\text{An}}) + \mu_{ij} (H_i - H_i^{\text{An}}) (H_j - H_j^{\text{An}}) \right) d\Omega} \quad (73)$$

where the expressions of  $S_{ij}^{\text{An}}$ ,  $E_i^{\text{An}}$  and  $H_i^{\text{An}}$  are given as

$$S_{ij}^{\text{An}} = \lim_{N \rightarrow \infty} \sum_{i=1}^N (\bar{B}_u^i)_{ijk} u_k, E_i^{\text{An}} = \lim_{N \rightarrow \infty} \sum_{i=1}^N (\bar{B}_\Phi^i)_{ij} \Phi, H_i^{\text{An}} = \lim_{N \rightarrow \infty} \sum_{i=1}^N (\bar{B}_\Psi^i)_{ij} \Psi \quad (74)$$

Demonstrate the efficiency of ISNS-RPIM through the parameter  $EC$

$$z_{ir} = z + r_A \cdot \Delta z \cdot \alpha_{ir} \quad (77)$$

where  $\alpha_{ir}$  represents the irregular factor whose value range is 0 to 0.5,  $r_A$  is the arbitrary number,  $\Delta x$  and  $\Delta z$  express the distances of regular nodes in the  $x$ -direction and  $z$ -direction.

The constraints corresponding to different boundary conditions are:  
Free end:  $u_x = u_z = \Phi = \Psi \neq 0$ .

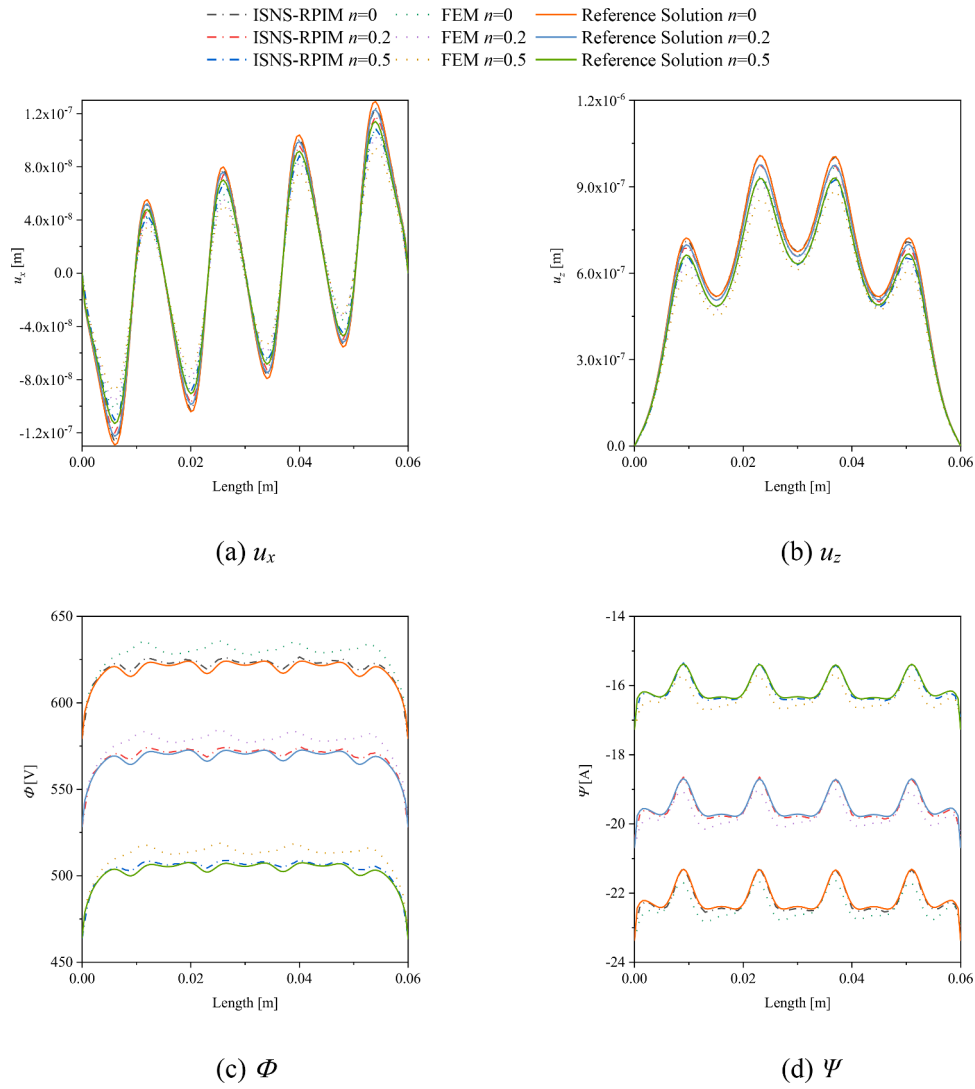


Fig. 21. The generalized displacement on edge AB ( $\alpha^* = \beta^* = -0.05$ ).

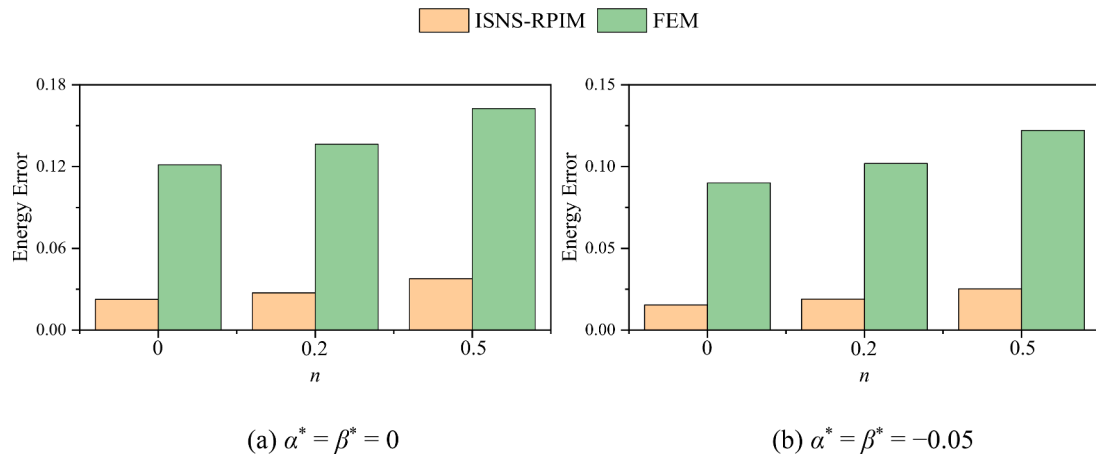


Fig. 22. Energy error.

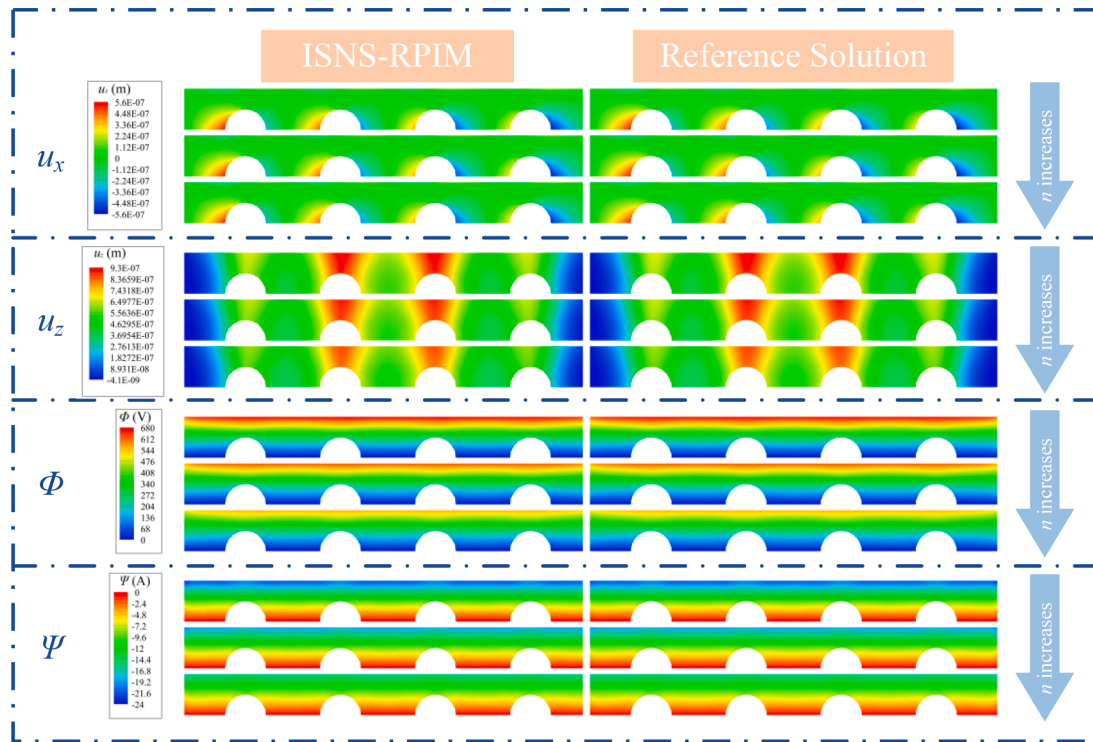


Fig. 23. FGME energy harvester Contours ( $\alpha^* = \beta^* = 0$ ).

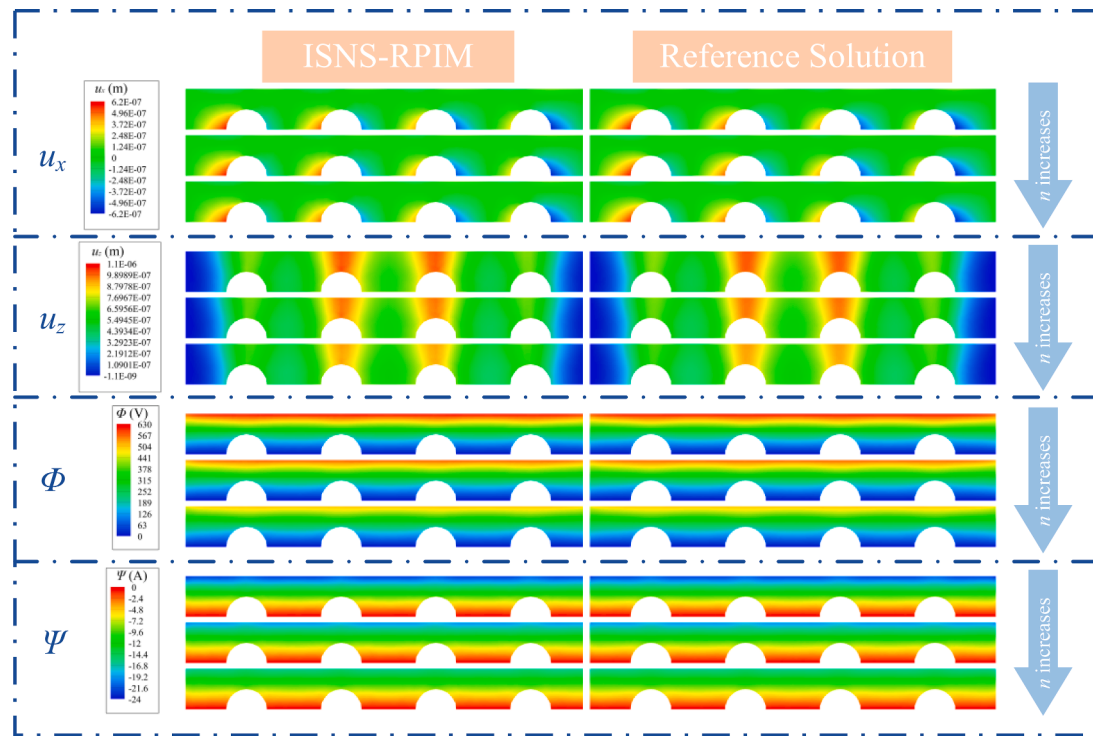


Fig. 24. FGME energy harvester contours ( $\alpha^* = \beta^* = -0.05$ ).

Bottom edge:  $\Phi = \Psi = 0$ .  
Clamped end:  $u_x = u_z = 0$ .

#### 4.1. MEE plate

The developed method is used to study the generalized displacement for a  $\text{BaTiO}_3\text{-CoFe}_2\text{O}_4$  in-plane plate to verify the correctness of ISNS-RPIM. Fig. 4 shows the meshes and dimensions of the structure

subjected to a uniform load  $F$  (upward) at the upper edge,  $F = 100 \text{ N/m}$ . The exponential factor is 0. The mechanical, electric and magnetic boundary conditions considered in the present work are:

$$u_x = 0 \text{ at } z = -1 \text{ (AD)}$$

$$u_z = \Phi = \Psi = 0 \text{ at } x = -1 \text{ (CD)}$$

Results at point  $B$  (1m, 1m) are given in Table 2. It can be observed that ISNS-RPIM displays a close agreement with the results of Zhu et al. [63]. Hence, ISNS-RPIM could produce sufficiently reliable results for



HME coupling problems. The formulation could be extended to analyze other structures with complex geometries, boundary conditions and loadings.

In the subsequent section, select the calculation results of FEM as the reference solution to further prove the advantages of ISNS-RPIM in solving HME coupling problems.

Table 3 shows the comparison of CPU time and error under different numerical methods. We compared the conclusion of ISNS-RPIM program with an existing reference solution element free Galerkin method (EFGM) [64]. It can be seen from Table 3 that the calculation efficiency of ISNS-RPIM is higher than FEM and EFGM, which proves the accuracy and high efficiency of ISNS-RPIM.

#### 4.2. FGME beam

The static responses and free vibration for FGME clamped-clamped beam subject to hygrothermal loadings are researched in this section. The beam with length  $L = 0.1$  m and width  $W = 0.01$  m is shown in Fig. 5. The empirical constants are  $\alpha^* = \beta^* = -0.01$ .

##### 4.2.1. Static and dynamic analysis for FGME beam

The exponential factor  $n$  is taken as 0, 0.2, 0.5. ISNS-RPIM uses triangular elements (427 nodes), and the calculation results as a reference using FEM based on quadrilateral elements (2737 nodes). Fig. 6 shows the generalized displacement on edge AB of the geometry, and Fig. 7 exhibits the contours of the whole mode. The variation of temperature and moisture are  $\Delta T = 50$  K and  $\Delta m = 1\%$ , respectively. It can be observed that under the same hygrothermal loading and  $n$ , with the increase of  $n$ , values of  $u_x$  and  $\Phi$  decrease and values of  $u_z$  and  $\Psi$  increases.

The same meshes are applied to solve the free vibration problem. Results are given in Fig. 8, the loadings are  $\Delta T = 10$  K,  $\Delta m = 1\%$  and  $\Delta T = 20$  K,  $\Delta m = 2\%$  and  $\Delta T = 50$  K,  $\Delta m = 3\%$ , the exponential factors are  $n = 0, 0.2, 0.5$ . It is obvious that, because of the existence of empirical constants affects the elastic stiffness coefficient, the natural frequency of the model decreases when the loading gradually increases. Where  $\alpha^*$  and  $\beta^*$  are negative, the elastic stiffness coefficient and the model stiffness decrease due to the increase in  $\Delta T$  and  $\Delta m$ . Fig. 9 shows the relative error values of the free vibration problem between different algorithms and reference solutions. The relative error of ISNS-RPIM is smaller than other algorithms including NS-RPIM and FEM. In addition, the figures reveal that NS-RPIM is inaccurate in solving the free vibration problem due to the temporal instability, which is far from the reference solution. Therefore, ISNS-RPIM has extremely high accuracy when solving statics and free vibration problems of FGME structures.

##### 4.2.2. Energy error and efficiency study

The energy error, CPU time and efficiency are shown in Fig. 10 with different algorithms including ISNS-RPIM, NS-RPIM and FEM.  $\Delta T = 10$  K,  $\Delta m = 1\%$  and  $\Delta T = 20$  K,  $\Delta m = 2\%$  are the hygrothermal loadings,  $n = 0, 0.2, 0.5$  are the exponential factors. 63, 205, 427 and 729 nodes were used for calculation. The expression of  $h$  is  $h = \sqrt{4\bar{A}/\sqrt{3}}$ , which represents the characteristic length.  $\bar{A}$  is the mean size of the element. The smaller the  $h$ , the more nodes are required for the computation. It is evident that the error decreases as  $h$  decreases, which illustrates the convergence of ISNS-RPIM. The errors of ISNS-RPIM are the smallest under any loadings and exponential factor. In terms of CPU time, NS-RPIM has longer calculation time than FEM due to the smoothing operation. ISNS-RPIM is based on NS-RPIM, and it also needs to calculate the stable terms, thus ISNS-RPIM takes the longest time to calculate the same problem.

EC is an indicator describing the efficiency of the algorithm. ISNS-RPIM with 63 nodes is not the most efficient one. However, in other cases, ISNS-RPIM has higher efficiency. In process of solving practical engineering problems, the requirements for accuracy and efficiency are

very high. ISNS-RPIM could combine high accuracy with high efficiency. Therefore, ISNS-RPIM is an algorithm suitable for solving practical HME coupling problems.

##### 4.2.3. Mesh distortion effect

The influence of irregular factor  $\alpha_{ir}$  on ISNS-RPIM is researched. Ten irregular factors are considered,  $\alpha_{ir} = 0.05, 0.1, 0.15, 0.2, 0.25, 0.3, 0.35, 0.4, 0.45, 0.5$ , the larger the  $\alpha_{ir}$ , the more irregular the meshes. Run the process ten times to obtain ten sets of generalized displacement. Then take the average of these ten groups of results and draw the curves on edge AB (Fig. 11) and calculate the errors (Fig. 12). A total of 245 nodes are used for calculation, two sets of hygrothermal loadings are considered,  $\Delta T = 10$  K,  $\Delta m = 1\%$  and  $\Delta T = 20$  K,  $\Delta m = 2\%$ . When  $\alpha_{ir} > 0.3$ , Jacobian matrix could obtain negative determination, which reduces the accuracy of FEM [46]. It can be seen from the results that ISNS-RPIM is the algorithm with the best agreement with the reference solution, which possesses the lowest error. There are no Jacobian matrix and coordinate transformation needed in the calculation process of ISNS-RPIM, thus, the ISNS-RPIM is insensitive to distorted mesh.

#### 4.3. Static responses of FGME sensor using ISNS-RPIM

After verifying the superiority of ISNS-RPIM, the HME coupling problems of complex FGME structures are analyzed, empirical constants and exponential factor are also taken into consideration to explore the impact on the results.

##### 4.3.1. Influence of $\Delta T$ and $\Delta m$

In Fig. 13, consider a FGME sensor with fully clamped on left and right sides. ISNS-RPIM and FEM use triangular elements (570 nodes), and the reference solution uses quadrilateral meshes with a much denser mesh (2060 nodes). In this section, two hygrothermal loadings are considered:  $\Delta T = 10$  K,  $\Delta m = 1\%$  and  $\Delta T = 40$  K,  $\Delta m = 3\%$ . The empirical constants are  $\alpha^* = \beta^* = -0.02$ , Consider the case where the exponential factor  $n$  is 0, 0.2, 0.5.

Figs. 14 and 15 display the  $u_x, u_z, \Phi, \Psi$  on edge AB, Fig. 16. exhibits the energy errors. The contours of the FGME sensor under different hygrothermal loadings and exponential factors are shown in Figs. 17 and 18. It can be seen that when the values of  $\alpha^*, \beta^*$  and  $n$  remain unchanged, the increases in  $\Delta T$  and  $\Delta m$  result in the increase of the generalized displacement.  $\Delta T$  and  $\Delta m$  increase, the structural stiffness decreases. In addition,  $\Phi$  and  $\Psi$  are greatly affected by  $n$ . As  $n$  increases, the absolute values of  $u_x, \Phi$  and  $\Psi$  decrease, those of  $u_z$  increases. According to Eq. (5), it can be seen that when  $n > 0$ , the stiffness of the structure increases, when the hygrothermal loading is kept constant, the displacement decreases. The form of the structure has an effect on the local displacement. The form of the structure has an effect on the local displacement.

It is clear that ISNS-RPIM is the most accurate one, whose results fit well with reference solutions, which proves again the correctness of ISNS-RPIM.

##### 4.3.2. Effective of empirical constants $\alpha^*$ and $\beta^*$

Next, we consider a FGME energy harvester with four semicircular holes at the bottom edge, left and right edges are both clamped in Fig. 19. In this section,  $\Delta T = 10$  K,  $\Delta m = 1\%$ .  $\alpha^* = \beta^* = 0$  and  $\alpha^* = \beta^* = -0.05$  are the empirical constants,  $n = 0, 0.2, 0.5$  are the exponential factor. ISNS-RPIM and FEM use triangular elements (416 nodes), the results as a reference based on quadrilateral meshes with a much denser mesh (1476 nodes).

Figs. 20 and 21 exhibit the generalized displacement on edge AB. Fig. 22 is the energy error of two algorithms. The contours of the FGME energy harvester are given in Figs. 23 and 24. When  $\Delta T, \Delta m$  and  $n$  do not change, the decrease of empirical constants leads to the decrease of structural stiffness. Therefore, values of  $u_x$  and  $u_z$  both increase, which is consistent with Eq. (50). Meanwhile,  $\Phi$  and  $\Psi$  decrease. It can also be

seen that  $\Phi$  and  $\Psi$  are greatly affected by  $n$ . As  $n$  increases, the value of  $\Phi$  decreases and the value of  $\Psi$  increases. The displacement is less affected by  $n$ , which decreases as  $n$  increases.

It is obvious that the error of ISNS-RPIM is the lowest, indicating that the developed method is feasible and valid in tackling complex FGME structures in hygrothermal environment.

## 5. Conclusion

The ISNS-RPIM is presented to explore FGME structures. The stable terms related to field variables are employed to improve the temporal stability of NS-RPIM. The factors influencing structural static responses and natural frequency are investigated in detail: exponential factor and hygrothermal loadings. Results show that increasing the empirical constants decreased the structural stiffness and the natural frequency. For exponential factors,  $u_x$ ,  $\Phi$  decrease and  $\Psi$  increases, as the exponential factor increase. In addition,  $\Phi$  and  $\Psi$  are greatly affected by the exponential factor. Finally, the results also show the advantages of high precision, high effectiveness, and insensitivity to mesh distortion in ISNS-RPIM, which verifies the robustness of ISNS-RPIM and shows the potential of the algorithm in solving practical complex problems. (Eqs. (1)–(4), (8), (10)–(12), (17)–(21), (23), (25)–(48), (52)–(77))

## Declaration of Competing Interest

The authors declare that they have no conflicts of interest.

## Data availability

No data was used for the research described in the article.

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